

Incipient Voltage Sensor Fault Isolation for Rectifier in Railway Electrical Traction Systems

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Abstract—This paper proposes a dc voltage incipient sensor fault isolation method for single-phase three-level rectifier devices in high-speed railway electrical traction systems. Different incipient fault modes characterizing locations and incipient fault types are parameterized nonlinearly by unknown fault parameters. A new incipient fault isolation method is developed by combining sliding mode technique with nonlinear parameterization adaptive estimation technique. A bank of particular adaptive sliding mode estimators is proposed, which facilitates to derive new isolation residuals and adaptive threshold intervals. The isolability is studied, and the isolable sufficient condition is derived using new functions. For the practical electrical traction system in CRH2 (China Railway High-Speed 2), simulation and experiment based on TDCS-FIB (a software) are presented to verify the effectiveness and feasibility of the proposed method.

Index Terms—Rectifiers dc voltage sensors, fault isolation, sliding mode and adaptive techniques.

I. INTRODUCTION

CHINA high-speed railways are fed by $2 \times 25\text{KV}/50\text{Hz}$ single phase ac current sources [1]. The ac/dc/ac electrical traction system in CRH2 includes a single-phase three-level ac-dc rectifier, a three-phase dc-ac inverter and four driving motors where the rectifier is different from the two-level rectifier discussed in [2]. Such a power electronic system may experience static electricity corrosion, high humidity and high temperature, electrical loading and mechanical vibration. For example, the electrical traction drive for an urban tram may experience $10^6 - 10^8$ power cycles in its lifetime, with temperature swings up to 80°C [3]. In addition, aging components in sensors, such as electrolyte loss in electrolytic capacitors which are most fragile [4], may result in incipient faults and further develop into serious failures. Therefore, early incipient sensor fault diagnosis is essential and significant.

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During the past decades, model-based fault diagnosis has been widely studied and applied such as [5] and [6]. The sliding mode observer based FDI (fault detection and isolation) has been extensively studied [7], [8] and [9]. In [7] and [8], the “equivalent output injection” concept is used to explicitly reconstruct fault signals to detect and isolate sensor faults and actuator faults. Using a sliding mode observer, an actuator FDI scheme is developed by generating residuals instead of reconstructing fault signals in [9]. All the above sliding mode observer based methods require that the fault signals are bounded by known functions. In addition, the challenging problem of residuals’ convergence for mismatched fault modes is not considered in [9]. Adaptive is a significant technique to solve these problems, especially to deal with uncertainties such as [10] and [11], and adaptive observer based FDI has been studied in [12], [13], [14], etc.. However, residuals and adaptive thresholds generated in these papers are conservative. Sliding mode technique in combination with adaptive technique is a pertinent solution to improve robustness of FDI against modeling uncertainties and disturbances. In [15] and [16], the adaptive and sliding mode techniques are used to estimate fault signals. However, residual based fault isolation by combining adaptive approach together with sliding mode technique is rarely few. In practical rectifiers, there always exist switching noises and grid-side voltage and current harmonics. However, sliding mode techniques have good robustness and are completely insensitive to the so-called matched uncertainty, and can also be used to deal with both structural and unmatched uncertainty (see e.g. [17]). They are usually used to control and monitor electrical equipments such as rectifiers, inverters and induction motors [18]. Therefore, the application of sliding mode techniques for rectifier incipient voltage sensor fault isolation offers good potential.

In this paper, a multi incipient sensor faults isolation method is developed for single-phase three-level rectifier dc voltage sensors in high-speed railway electrical traction systems. The voltage incipient sensor faults are modeled by differential equations with unknown inputs parameterized by fault parameters in a nonlinear way. To isolate different fault modes, a bank of particular designed sliding mode and adaptive estimators is proposed. In the FIEs (fault isolation estimators), the projection adaptive laws are proposed based on Min-Max method proposed in [19] to estimate the fault parameters. In addition, all the estimation errors generated by FIEs are ensured to enter into the sliding surface in finite time and maintain on it thereafter, no matter whether the matched or mismatched fault modes occur. New residuals are defined,

and adaptive threshold intervals are designed based on the interval observer theory (see e.g., [20]). The incipient sensor fault mode isolable condition characterizing which fault mode can be isolated, is derived by new defined functions. The effectiveness and feasibility of the proposed dc voltage incipient sensor fault isolation method is tested using the TDCS-FIB program provided by Central South University fault injection team.

Notations: For a real matrix or a vector M , $M > 0$ ($M \geq 0$) means that its entries are positive (nonnegative). The symbol $\text{diag}(v)$ denotes a diagonal matrix with the diagonal elements formed by the elements of the vector v . The identity matrix with dimension n is denoted by I_n . The n dimensional column vector with all elements being 1 is denoted by \mathcal{I}_n .

II. MODELING AND PRELIMINARIES

A. Rectifier Modeling

The equivalent schematic diagram of the single-phase three-level PWM rectifier device in CRH2 is shown in Fig. 1 where u_s and i_s are the grid side voltage and current respectively, S_{ij} , $i = 1, 2, 3, 4$, $j = a, b$ are the IGBT modules of bridges a and b , respectively, u_1 and u_2 are the voltages of capacities C_1 and C_2 in dc-link side respectively, and i_l is the load current. For simplifying analysis, the ideal switching functions S_j , $j = a, b$ are defined as in [21]. Suppose that there is no power loss and

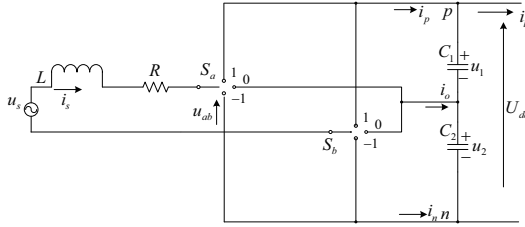


Fig. 1. Switch equivalent circuit

energy storage in IGBTs. Then it yields that

$$u_{ab} = S_1 u_1 - S_2 u_2, \quad i_p = S_1 i_s \quad \text{and} \quad i_n = S_2 i_s. \quad (1)$$

where $S_1 = \frac{S_a(S_a+1) - S_b(S_b+1)}{2}$ and $S_2 = \frac{S_a(S_a-1) - S_b(S_b-1)}{2}$. Suppose that there is also no power loss and energy storage in the ac/dc/ac electrical traction system. Then the instantaneous powers are equal between the dc side and traction motors, that is

$$(u_1 + u_2)i_l = P_m \quad (2)$$

where P_m is the instantaneous power of traction motors.

It should be pointed out that the IGBT switching and the harmonics in grid side voltage u_s result in harmonics in grid current i_s and dc voltages u_1 and u_2 . In this paper, these harmonics are considered as disturbances and uncertainties in the rectifier devices, and are represented by $\eta_n(u_s, i_s, u_1, u_2, \omega, t) = \text{col}(\eta_{i_s}(\cdot), \eta_{u_1}(\cdot), \eta_{u_2}(\cdot))$, which should be considered in fault diagnosis. From Fig. 1, on the

basis of the Kirchhoff current and voltage principles, it has that

$$\frac{di_s}{dt} = \frac{1}{L}(u_s - Ri_s - u_{ab}) + \eta_{i_s}(u_s, i_s, u_1, u_2, \omega), \quad (3)$$

$$\frac{du_1}{dt} = \frac{1}{C_1}(i_p - i_l) + \eta_{u_1}(u_s, i_s, u_1, u_2, \omega), \quad (4)$$

$$\frac{du_2}{dt} = \frac{1}{C_2}(-i_n - i_l) + \eta_{u_2}(u_s, i_s, u_1, u_2, \omega). \quad (5)$$

Thus, the state-space of the single-phase three-level rectifier device is obtained by

$$\dot{X} = A_n X + g_n(X) + B_n U + \eta_n(X, U, \omega, t) \quad (6)$$

$$y = X \quad (7)$$

where $X = \text{col}(i_s, u_1, u_2)$, $U = u_s$ and $B_n = \text{col}(1/L, 0, 0)$,

$$A_n = \begin{bmatrix} -\frac{R}{L} & -\frac{S_1}{L} & \frac{S_2}{L} \\ \frac{S_1}{C_1} & 0 & 0 \\ -\frac{S_2}{C_2} & 0 & 0 \end{bmatrix} \quad \text{and} \quad g_n(X) = \begin{bmatrix} 0 \\ -\frac{P_m}{C_1(X_2 + X_3)} \\ -\frac{P_m}{C_2(X_2 + X_3)} \end{bmatrix}.$$

B. Incipient Sensor Fault Modeling

An incipient sensor fault always develops in a continuous way that has been discussed in [22], which can be modeled based on the following lemma.

Lemma 1. [23] For any piecewise continuous vector function $f : \mathcal{R}^+ \rightarrow \mathcal{R}^q$, and a stable $q \times q$ matrix A_f , there always exists an input vector $\xi \in \mathcal{R}^q$ such that $\dot{f} = A_f f + \xi$.

For the dc voltage sensors, there are three fault modes: only sensor u_1 has incipient fault, only sensor u_2 has incipient fault and both sensors u_1 and u_2 have incipient faults simultaneously. From Lemma 1, these fault modes are modeled by

$$\dot{f}^i = A_f^i f^i + D_2^i \Delta(u_1, u_2, U, \theta^i), \quad f^i(0) = 0, \quad i = 1, 2, 3 \quad (8)$$

where A_f^i are Hurwitz matrices. The functions $\Delta(u_1, u_2, U, \theta^i)$ are continuous with small amplitude, which are used to describe the incipient faults. They are parameterized by θ^i nonlinearly. The faults considered are different from the existing linear parameterized faults as in [14]. It should be noted that A_f^i are not design parameters. The unknown input distribution matrix D_2^i , $i = 1, 2, 3$ are given by

$$D_2^1 = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}, \quad D_2^2 = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}, \quad D_2^3 = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}. \quad (9)$$

Let p and N denote the number of sensors and number of possible fault modes in the rectifier device respectively. Then a set Σ_D related to considered fault modes is defined by

$$\Sigma_D = \{D_2^i, i = 1, \dots, N\}. \quad (10)$$

The functions $\Delta(\cdot, \theta^i) := \text{col}\{\delta(\cdot, \theta_1^i), \dots, \delta(\cdot, \theta_p^i)\}$, $i = 1, \dots, N$ where $\theta^i = \text{col}(\theta_1^i, \dots, \theta_p^i) \in \mathcal{R}^{ph}$ with $\theta_j^i \in \mathcal{R}^h$, $j = 1, \dots, p$ represent the unknown fault parameter vectors, and are assumed to belong to known hypercubes Θ^i , given by

$$\theta^i \in \Theta^i = \{\theta^i \in \mathcal{R}^{ph} \mid \underline{\theta}_j^i \leq \theta_j^i \leq \bar{\theta}_j^i, j = 1, \dots, ph\}. \quad (11)$$

where $\underline{\theta}_j^i$ and $\bar{\theta}_j^i$ are upper bound and lower bound of θ_j^i respectively. Therefore, the matrix A_f^i and the unknown input

$D_2^i \Delta(\cdot, \theta^i)$ characterize a class of incipient fault modes, in which the structure and time varying “magnitude” are parameterized by A_f^i , D_2^i and θ^i .

Remark 1. In [14], it is assumed that only one sensor fault occurs, which is can be applied to the case when multi faults occur simultaneously in practical systems. In this paper, multi incipient sensor faults isolation schemes are developed. The set Σ_D in (10) contains multi incipient sensor faults mode in rectifier devices which will be isolated under certain conditions in this paper. ∇

C. Augmented System

Let $x^i := \text{col}(X, f^i) \in \mathcal{R}^{n+p}$ where n is the dimension of the rectifier. Inverter model (6)-(7) and fault models (8) can be represented in an augmented form as

$$\dot{x}^i = A^i x^i + g(x^i) + Bu + \eta(x^i, u, \omega, t) + D^i \Delta(y, u, \theta^i), \quad (12)$$

$$y = [I_p, I_p] x^i, \quad i = 1, \dots, N \quad (13)$$

where y is the actual measured signals, $u = U$, $A^i = \text{diag}(A_n, A_f^i)$, $g(x^i) = \text{col}(g_n(C_x x^i), 0) = \text{col}(g_{n0}(x^i), 0)$ with $C_x = \text{diag}(I_n, 0)$, $B = \text{col}(B_n, 0)$, $\eta(\cdot) = \text{col}(\eta_n(\cdot), 0)$ and $D^i = \text{col}(0, D_2^i)$. It should be pointed out that D^i are not necessary full column rank, which are different from traditional papers such as in [7] and [8].

It is assumed throughout this paper that $\text{rank}(D_2^i) = q_i$, $i = 1, \dots, N$, which mean that there are q_i elements 1 in diagonal of D_2^i . Since D_2^i and D_2^j ($j \neq i$) are diagonal matrices with elements 1 and 0 in diagonal, there exists one and only one orthogonal matrix $T_0^i \in \mathcal{R}^{p \times p}$ such that

$$(T_0^i)^T D_2^i T_0^i = \left[\begin{array}{c|c} 0 & 0 \\ \hline 0 & D_{23}^{ii} \end{array} \right] \quad (14)$$

where $D_{23}^{ii} = I_{q_i}$. Then

$$(T_0^i)^T D_2^j T_0^i = \left[\begin{array}{c|c} D_{22}^{ij} & 0 \\ \hline 0 & D_{23}^{ij} \end{array} \right] \quad (15)$$

where $D_{22}^{ij} \in \mathcal{R}^{(p-q_i) \times (p-q_i)}$ and $D_{23}^{ij} \in \mathcal{R}^{q_i \times q_i}$ are diagonal matrices with only elements 1 and 0 in diagonals. Supposing that there are d_1^j elements 1 in D_{22}^{ij} and d_2^j elements 1 in D_{23}^{ij} , respectively, $d_1^j + d_2^j = q_j$. Let

$$T_a^i = \left[\begin{array}{c|c} I_n & 0 \\ \hline I_p & (T_0^i)^T \end{array} \right], \quad T_b^i = T_0^i. \quad (16)$$

Then

1) $z^i = T_a^i x^i = \text{col}(z_1, z_2^i) = \text{col}(X, X + (T_0^i)^T f^i)$ where $z_1 \in \mathcal{R}^n$ and $z_2^i \in \mathcal{R}^p$,

2) $T_a^i A^i (T_a^i)^{-1} = \left[\begin{array}{c|c} A_{11} & 0 \\ \hline A_{21}^i & A_{22}^i \end{array} \right] = \left[\begin{array}{c|c} A_{11} & 0 \\ \hline \frac{A_{21}^i}{A_{22}^i} & \frac{A_{22}^{i11}}{A_{22}^i} \quad \frac{A_{22}^{i12}}{A_{22}^i} \end{array} \right] =$

$\left[\begin{array}{c|c} A_n & 0 \\ \hline A_n - A_f^i & T_0^i A_f^i (T_0^i)^T \end{array} \right]$ where $A_{11} = A_n$, $A_{21}^i \in \mathcal{R}^{(p-q_i) \times n}$, $A_{22}^i \in \mathcal{R}^{q_i \times n}$, $A_{22}^{i11} \in \mathcal{R}^{(p-q_i) \times (p-q_i)}$ and $A_{22}^{i22} \in \mathcal{R}^{q_i \times q_i}$,

3) $T_a^i D^i T_b^i = \left[\begin{array}{c|c} 0 & \\ \hline 0 & 0 \\ \hline 0 & D_{23}^{ii} \end{array} \right]$, $T_a^i D^j T_b^i = \left[\begin{array}{c|c} 0 & \\ \hline D_{23}^{ij} & 0 \\ \hline 0 & D_{23}^{ij} \end{array} \right]$,

4) $[I_p, F](T_a^i)^{-1} = \left[\begin{array}{c} 0 \\ T_0^i \end{array} \right]$.

Accordingly, $T_a^i g(\cdot)$, $T_a^i \eta(\cdot)$ and $T_a^i B$ can be obtained.

With matrix $(T_b^i)^{-1}$, the j th $\Delta(\cdot, \theta^j)$, $j = 1, \dots, N$ is assumed to be reordered as $\Delta^i(\cdot, \theta^{ij}) = (T_b^i)^{-1} \Delta(\cdot, \theta^j)$ where

$$\Delta^i(\cdot, \theta^{ij}) = \text{col}(\Delta_1^i(\cdot, \theta^{ij1}), \Delta_2^i(\cdot, \theta^{ij2})), \quad (17)$$

$$\theta^{ij} = \text{col}(\theta^{ij1}, \theta^{ij2}) \quad (18)$$

where

$$\Delta_1^i(\cdot, \theta^{ij1}) = \text{col}(\delta(\cdot, \theta_1^{ij}), \dots, \delta(\cdot, \theta_{p-q_j}^{ij})), \quad (19)$$

$$\theta^{ij1} = \text{col}(\theta_1^{ij}, \dots, \theta_{p-q_j}^{ij}) \in \mathcal{R}^{(p-q_j)h}, \quad (20)$$

$$\Delta_2^i(\cdot, \theta^{ij2}) = \text{col}(\delta(\cdot, \theta_{p-q_j+1}^{ij}), \dots, \delta(\cdot, \theta_p^{ij})), \quad (21)$$

$$\theta^{ij2} = \text{col}(\theta_{p-q_j+1}^{ij}, \dots, \theta_p^{ij}) \in \mathcal{R}^{q_j h}. \quad (22)$$

Furthermore, for the j th fault mode, the unknown fault parameter vector θ^{ij1} in (20) and θ^{ij2} in (22) are assumed to belong to known hypercubes Θ^{ij1} and Θ^{ij2} respectively, which can be obtained based on (11) and the matrix T_b^i .

Remark 2. It should be pointed out that different fault modes j and transformations T_b^i have different Θ^{ij1} and Θ^{ij2} , which provide the capabilities to isolate different fault modes. ∇

Assumption 1. The modeling uncertainties $\eta(\cdot)$ in (12) satisfies that $\forall t > 0$, $\|\eta(\cdot)\| \leq \bar{\eta}$ where $\bar{\eta}$ is a known constant. Moreover, the nonlinear function $g_{n0}(x^i)$ in (12) is uniformly Lipschitz, i.e. $\|g_{n0}(x^i) - g_{n0}(\hat{x}^i)\| \leq \mathcal{L} \|x^i - \hat{x}^i\|$ where \mathcal{L} is the known Lipschitz constant.

Remark 3. In this paper, there is no constraint on the structure of uncertainties. The voltage and current harmonics are main sources of disturbances, and uncertain capacitances and inductances are the main sources of uncertainties in practical rectifier systems. In general, the injection expressions η_{i_s} , η_{u_1} and η_{u_2} in system (3)-(5) caused by harmonics and uncertain capacitances and inductances can be obtained based on rectifier working mechanism. Then the bounds $\bar{\eta}$ on uncertainties $\eta(\cdot)$ in (12) can be obtained based on specifications provided by manufacturers. Moreover, the historic statistical data can be used to help to estimate the bounds on external disturbances. ∇

For the i th fault mode, augmented system (12)-(13) is transformed to

$$\dot{z}_1 = A_n z_1 + g_{n0}((T_a^i)^{-1} z^i) + B_n u + \eta_n(\cdot), \quad (23)$$

$$\dot{z}_{21}^i = A_{211}^i z_1 + A_{22}^{i11} z_{21}^i + A_{22}^{i12} z_{22}^i + g_{21}^i((T_a^i)^{-1} z^i) + B_{21}^i u + \eta_{21}^i(\cdot), \quad (24)$$

$$\dot{z}_{22}^i = A_{212}^i z_1 + A_{22}^{i21} z_{21}^i + A_{22}^{i22} z_{22}^i + g_{22}^i((T_a^i)^{-1} z^i) + B_{22}^i u + \eta_{22}^i(\cdot) + D_{23}^{ii} \Delta_2^i(y, u, \theta^{ii2}), \quad (25)$$

$$y = T_0^i z_2^i \quad (26)$$

where $z_2^i = \text{col}(z_{21}^i, z_{22}^i)$ with $z_{21}^i = [I_{p-q_i}, 0](T_0^i)^T y \in \mathcal{R}^{p-q_i}$, $z_{22}^i = [0, I_{q_i}](T_0^i)^T y \in \mathcal{R}^{q_i}$. In (24) and (25), the functions g_{21}^i and g_{22}^i , the matrices B_{21}^i and B_{22}^i , and the uncertainties η_{21}^i and η_{22}^i are obtained from T_a^i .

Fault detection is to detect the occurrence of faults, while fault isolation focuses on localization (classification) of different fault modes. In this paper, multi incipient sensor faults isolation schemes will be developed based on sliding mode technique and nonlinear parameterization adaptive estimation technique, which are completely different from [22].

III. INCIPIENT SENSOR FAULT ISOLATION SCHEMES

A. FIEs Design and Incipient Fault Isolation Decision Scheme

Different from FIEs in [13], the following FIEs are constructed by combining sliding mode technique with nonlinear parameterization adaptive estimation technique. Suppose that the incipient sensor fault is detected at time T_d when FIEs are activated. Each FIE corresponds to one potential fault mode. There are N FIEs corresponding to N potential fault modes. Denote \hat{z}_1 , \hat{z}_{21}^s and \hat{z}_{22}^s as estimation of z_1 , z_{21}^s and z_{22}^s respectively, and $e_{22}^s = z_{22}^s - \hat{z}_{22}^s$. The s th FIE is designed as

$$\dot{\hat{z}}_1 = A_n \hat{z}_1 + g_{n0}((T_a^s)^{-1} \hat{z}^s) + B_n u, \quad \hat{z}_1(T_d) = 0, \quad (27)$$

$$\begin{aligned} \dot{\hat{z}}_{21}^s &= A_{211}^s \hat{z}_1 + A_{22}^{s11} \hat{z}_{21}^s + A_{22}^{s12} z_{22}^s + g_{21}^s((T_a^s)^{-1} \hat{z}^s) \\ &\quad + K_{11}^s (z_{21}^s - \hat{z}_{21}^s) + B_{21}^s u + \bar{D}_{22}^{ss} \Delta_1^s(\cdot, \hat{\theta}^{ss1}) \\ &\quad + \nu_{21}^s + \nu_{22}^s, \quad \hat{z}_{21}^s(T_d) = 0, \end{aligned} \quad (28)$$

$$\begin{aligned} \dot{\hat{z}}_{22}^s &= A_{212}^s \hat{z}_1 + A_{22}^{s21} \hat{z}_{21}^s + A_{22}^{s22} \hat{z}_{22}^s + K_{22}^s (z_{22}^s - \hat{z}_{22}^s) \\ &\quad + \hat{A}_{22}^{s22} \text{sat}\left(\frac{e_{22}^s}{\epsilon}\right) + g_{22}^s((T_a^s)^{-1} \hat{z}^s) + B_{22}^s u \\ &\quad + D_{23}^{ss} \Delta_2^s(\cdot, \hat{\theta}^{ss2}) + D_{23}^{ss} \hat{A}_2^s(\cdot, \hat{\theta}^{ss2}) \text{sat}\left(\frac{e_{22}^s}{\epsilon}\right) \\ &\quad - \Psi^s(z_{21}^s, z_{22}^s), \quad \hat{z}_{22}^s(T_d) = 0 \end{aligned} \quad (29)$$

where $\hat{z}^s \triangleq \text{col}(\hat{z}_1, (T_0^s)^T y)$, the matrix $K_{11}^s = A_{22}^{s11} - \hat{A}_{22}^{s11}$ with \hat{A}_{22}^{s11} being symmetric negative definite, the matrix $\bar{D}_{22}^{ss} = I_{p-q_s}$. The matrix $K_{22}^s = A_{22}^{s22} - \hat{A}_{22}^{s22}$ with \hat{A}_{22}^{s22} being symmetric negative definite and Metzler* where the positive system theory is used. The positive constant ϵ is chosen as a small scalar. The vector function $\hat{\Lambda}_2^s(\cdot) = \text{diag}(\Lambda_2^s(\cdot))$ where $\Lambda_2^s(\cdot)$ is given in (74) with α being chosen as e_{22}^s (see Appendix). $\Psi^s(\cdot) \in \mathcal{R}^{q_s}$ will be determined later. The functions ν_{21}^s and ν_{22}^s are defined by

$$\nu_{21}^s = m_{21}^s(\cdot) \text{sgn}(z_{21}^s - \hat{z}_{21}^s), \quad \nu_{22}^s = M_{22}^s(\cdot) \text{sgn}(z_{22}^s - \hat{z}_{22}^s) \quad (30)$$

where the scalar function $m_{21}^s(\cdot)$ and the diagonal matrix $M_{22}^s(\cdot)$ are determined later.

The vectors $\hat{\theta}^{ss1}$ and $\hat{\theta}^{ss2}$ are estimations of θ^{ss1} and θ^{ss2} respectively. The update laws, derived using Min-Max approach (see e.g., [19] and [24]), are proposed by

$$\dot{\hat{\theta}}^{ss1} = \mathcal{P}_{\Theta^{s1}} \left\{ -\Gamma_{\theta^{s1}} e_{21}^s W_1^s(\cdot, \hat{\theta}^{ss1}) \right\}, \quad \hat{\theta}^{ss1}(0) \in \Theta^{s1}, \quad (31)$$

$$\dot{\hat{\theta}}^{ss2} = \mathcal{P}_{\Theta^{s2}} \left\{ -\Gamma_{\theta^{s2}} e_{22}^s W_2^s(\cdot, \hat{\theta}^{ss2}) \right\}, \quad \hat{\theta}^{ss2}(0) \in \Theta^{s2} \quad (32)$$

where $\Gamma_{\theta^{s1}} = \Gamma_{\theta^{s1}}^T > 0$ and $\Gamma_{\theta^{s2}} = \Gamma_{\theta^{s2}}^T > 0$. The matrix function $W_1^s(\cdot)$ is given in (73) with α being chosen as e_{21}^s , and the matrix function $W_2^s(\cdot)$ is given in (74) with α being

*A real matrix is called Metzler matrix if all its off-diagonal entries are nonnegative.

chosen as e_{22}^s (See Appendix). In (31), the projection operator $\mathcal{P}_{\Theta^{s1}}$ restricts the parameter estimation $\hat{\theta}^{ss1}$ in Θ^{s1} . Also, $\mathcal{P}_{\Theta^{s2}}$ in (32) restricts the parameter estimation $\hat{\theta}^{ss2}$ in Θ^{s2} . In order to enter into sliding motion and guarantee the stability, the convex regions Θ^{s1} and Θ^{s2} are defined as

$$\Theta^{s1} = \bigcup_{i=1}^N \Theta^{si1}, \quad \Theta^{s2} = \Theta^{ss2}. \quad (33)$$

where Θ^{si1} and Θ^{ss2} are obtained based on (11) and T_b^s .

This paper is different from the fault isolation schemes presented in [14] in that the adaptive threshold interval concept will be introduced later. Considering this, the **incipient sensor fault mode isolation decision principle** in this paper is presented as follows: *if, for each $h \in \{1, \dots, N\} \setminus \{s\}$, there exist some finite time $t^h > T_d$ and some $j \in \{1, \dots, q_h\}$ such that $r_j^h(t) \notin [\underline{\zeta}_j^h, \bar{\zeta}_j^h]$, then the occurrence of the s th fault mode is concluded, where $r_j^h(t)$ represents the j th residual of the h th FIE, and $[\underline{\zeta}_j^h, \bar{\zeta}_j^h]$ is the corresponding adaptive threshold interval. The fault isolation time is defined as $T_{isol}^s \triangleq \max\{t^h, h \in \{1, \dots, N\} \setminus \{s\}\}$.*

B. Stability Analysis

Denote $e_1 = z_1 - \hat{z}_1$ and $e_{21}^s = z_{21}^s - \hat{z}_{21}^s$. Define the hyperplane sliding surfaces \mathcal{S}^s for the s th FIE as follows

$$\mathcal{S}^s = \{(e_1, e_{21}^s) \mid e_{21}^s = 0\}, \quad s = 1, \dots, N. \quad (34)$$

Note that

$$(T_a^s)^{-1}(z^s - \hat{z}^s) = \begin{bmatrix} I_n & 0 \\ -T_0^s & T_0^s \end{bmatrix} \begin{bmatrix} z_1 - \hat{z}_1 \\ 0 \end{bmatrix} = \begin{bmatrix} I_n \\ -T_0^s \end{bmatrix} (z_1 - \hat{z}_1). \quad (35)$$

From (23) and (27),

$$\dot{e}_1 = A_n e_1 + g_{n0}((T_a^s)^{-1} z^s) - g_{n0}((T_a^s)^{-1} \hat{z}^s) + \eta_n(\cdot). \quad (36)$$

Thus, it can be obtained that $e_1 = e^{A_n t} e_1(T_d) + \int_{T_d}^t e^{A_n(t-\tau)} (g_{n0}((T_a^s)^{-1} z^s) - g_{n0}((T_a^s)^{-1} \hat{z}^s) + \eta_n(\cdot)) d\tau$.

In rectifier control systems, the objective is to ensure unity power factor operation and dc link voltage regulation. The inner loop of the rectifier device controls the input current i_s using a classical PI controller, which is designed as a classical I-type system. When switching frequencies are high enough, the close-loop current in inner loop is first-order inertial element. The external loop of the rectifier device controls the dc link voltages u_1 and u_2 using a classical PI controller, which is designed as a classical II-type system. The closed-loop external loop closed-loop system is a minimum phase stable three-order damper system. Therefore, in rectifier control systems, the switching signals S_a and S_b are designed such that the switching system (6) is ISS (input-to-state stable) and A_n is Hurwitz. Thus, from [25], there exist positive constants k_0 and λ_0 such that $\|e^{A_n t}\| \leq k_0 e^{-\lambda_0 t}$, $t \geq T_d$.

Suppose that there exists ω_1 such that $\|e_1(T_d)\| = \|z_1(T_d)\| \leq \omega_1$. Then by applying the Bellman-Gronwall lemma, it follows from Assumption 1 and (35) that $\|e_1(t)\|$ satisfies that

$$\|e_1(t)\| \leq \chi(\cdot), \quad t \geq T_d \quad (37)$$

where $\chi(\cdot) \triangleq \chi(k_0, \bar{\eta}, \lambda_0, \mathcal{L}, T_0^s)$ which can be obtained from [25] and [13].

With the transformation matrices T_a^s and T_b^s , in the presence of the h th ($h \in \{1, \dots, N\}$) fault mode, the unknown input of z_{21}^s is described by $D_{22}^{sh} \Delta_1^s(y, u, \theta^{sh1})$. By comparing with (28), the error dynamic is obtained by

$$\begin{aligned} \dot{e}_{21}^s = & A_{211}^s e_1 + \hat{A}_{22}^{s11} e_{21}^s + g_{21}^s((T_a^s)^{-1} z^s) - g_{21}^s((T_a^s)^{-1} \hat{z}^s) \\ & - \nu_{21}^s + D_{22}^{sh} \Delta_1^s(\cdot, \theta^{sh1}) - \bar{D}_{22}^{ss} \Delta_1^s(\cdot, \hat{\theta}^{ss1}) \\ & + \eta_{21}(\cdot) - \nu_{22}^s. \end{aligned} \quad (38)$$

The following proposition is ready to be presented.

Proposition 1. Let $\kappa \geq \sup \|\theta^{sh1} - \hat{\theta}^{ss1}\|$ with $\theta^{sh1} \in \Theta^{sh1}$ and $\hat{\theta}^{ss1}$ updated by (31). In the presence of the h th fault mode, $h \in \{1, \dots, N\}$, if the scalar function $m_{21}^s(y, u, t)$ and the diagonal matrix function $M_{22}^s(y, u, \hat{\theta}^{ss1})$ in (30) satisfy

$$\begin{aligned} m_{21}^s(\cdot) \geq & (\|A_{211}^s\| + \mathcal{L}(1 + \|T_0^s\|)) \chi(\cdot) \\ & + \|W_1^s(\cdot, \hat{\theta}^{ss1})\| \kappa + \|\hat{\Lambda}_1^s(\cdot, \hat{\theta}^{ss1})\| \\ & + |\Delta_1^s(\cdot, \hat{\theta}^{ss1})| + \bar{\eta} + \varpi, \end{aligned} \quad (39)$$

$$M_{22}^s(\cdot) = \hat{\Lambda}_1^s(\cdot, \hat{\theta}^{ss1}), \quad t \geq T_d \quad (40)$$

where $\hat{\Lambda}_1^s(\cdot) = \text{diag}(\Lambda_1^s(\cdot), \Lambda_1^s(\cdot))$ and $W_1^s(\cdot)$ are given in (73) (see Appendix), then e_{21}^s will enter into the hyperplane \mathcal{S}^s given in (34) in finite time and will remain on it thereafter.

Proof: Let $V = (e_{21}^s)^T e_{21}^s$. It follows from (38) that $\dot{V} = \dot{V}_1 + \dot{V}_2 + \dot{V}_3$ where $\dot{V}_1 = (e_{21}^s)^T (\hat{A}_{22}^{s11} + (\hat{A}_{22}^{s11})^T) e_{21}^s < 0$ due to that \hat{A}_{22}^{s11} is symmetric negative definite,

$$\begin{aligned} \dot{V}_2 = & (e_{21}^s)^T \{A_{211}^s e_1 - D_{22}^{sh} W_1^s(\cdot, \hat{\theta}^{ss1})(\theta^{sh1} - \hat{\theta}^{ss1})\} \\ & + g_{21}^s((T_a^s)^{-1} z^s) - g_{21}^s((T_a^s)^{-1} \hat{z}^s) - \eta_{21}^s(\cdot) - \nu_{21}^s \end{aligned} \quad (41)$$

and

$$\begin{aligned} \dot{V}_3 = & (e_{21}^s)^T \{D_{22}^{sh} \Delta_1^s(\cdot, \theta^{sh1}) - \bar{D}_{22}^{ss} \Delta_1^s(\cdot, \hat{\theta}^{ss1}) \\ & + D_{22}^{sh} W_1^s(\cdot, \hat{\theta}^{ss1})(\theta^{sh1} - \hat{\theta}^{ss1}) - \nu_{22}^s\}. \end{aligned} \quad (42)$$

Note that $D_{22}^{sh} + \bar{D}_{22}^{sh} = I_{p-q_s} = \bar{D}_{22}^{ss}$. Then, $\dot{V}_3 = \dot{V}_{31} + \dot{V}_{32}$ where

$$\begin{aligned} \dot{V}_{31} = & (e_{21}^s)^T \{D_{22}^{sh} \Delta_1^s(\cdot, \theta^{sh1}) - D_{22}^{sh} \Delta_1^s(\cdot, \hat{\theta}^{ss1}) \\ & + D_{22}^{sh} W_1^s(\cdot, \hat{\theta}^{ss1})(\theta^{sh1} - \hat{\theta}^{ss1}) \\ & - D_{22}^{sh} \hat{\Lambda}_1^s(\cdot, \hat{\theta}^{ss1}) \text{sgn}(e_{21}^s)\}, \end{aligned} \quad (43)$$

$$\begin{aligned} \dot{V}_{32} = & -(e_{21}^s)^T \{\bar{D}_{22}^{ss} \Delta_1^s(\cdot, \hat{\theta}^{ss1}) \\ & + \bar{D}_{22}^{ss} \hat{\Lambda}_1^s(\cdot, \hat{\theta}^{ss1}) \text{sgn}(e_{21}^s)\}. \end{aligned} \quad (44)$$

From (69), (70) and (73), and choosing $\alpha = e_{21}^s$ in Appendix, it follows from the gain $M_{22}^s(\cdot)$ in (40) that $\Pi_i = |e_{21i}^s| (\text{sgn}(e_{21i}^s) (\delta(\cdot, \theta_i^h) - \delta(\cdot, \hat{\theta}_i^s) + w_i(\cdot) (\theta_i^h - \hat{\theta}_i^s)) - \lambda_i(\cdot)) \leq 0$. Then, $\dot{V}_{31} = \sum_{i=1}^{q_h} \Pi_i < 0$. Due to the fact that $\kappa \geq \sup \|\theta^{sh1} - \hat{\theta}^{ss1}\|$, $\|D_{22}^{sh}\| \leq 1$ and $\|\bar{D}_{22}^{sh}\| \leq 1$, and also $\|\eta_{21}\| < \bar{\eta}$, $\|g_{21}^s((T_a^s)^{-1} z^s) - g_{21}^s((T_a^s)^{-1} \hat{z}^s)\| \leq \mathcal{L}(1 + \|T_0^s\|) \|e_1\|$, the gain $m_{21}^s(\cdot)$ in (39) ensures that $\dot{V}_2 + \dot{V}_{32} \leq -\varpi \|e_{21}\|$. Therefore, $\dot{V} \leq -\varpi \|e_{21}\| \leq -\varpi V^{1/2}$, which implies that the reachability condition is satisfied.

Hence, the conclusion follows. \blacksquare

Remark 4. It should be pointed out that these isolation schemes require that all sliding motions take place earlier than

faults occur. Compared with abrupt faults, incipient sensor faults in rectifiers usually take long time to cause system failures. In addition, the time taken to reach sliding surfaces (34) can be reduced by adjusting ϖ and $\hat{z}_{21}^s(0)$ to ensure that the sliding motions occur at the very initial stage. Therefore, the developed results can be applied to a majority of cases in reality. ∇

C. Adaptive Threshold Interval

Based on the fault mode isolation principle presented in Section III-A, the s th fault isolation residual r^s is required to be sensitive not only to the s th fault mode, but also to the h th fault mode with $h \in \{1, \dots, N\} \setminus \{s\}$, which is different from [14] since sliding mode is introduced in FIEs (27)-(29). Therefore, a tuning error $e_{22\epsilon}^s = e_{22}^s - \epsilon \text{sat}(\frac{e_{22}^s}{\epsilon}) + T^s z_{21}^s$ is defined as the isolation residual, where the constant matrix $T^s \in \mathcal{R}^{q_s \times (p-q_s)}$ is chosen such that $T^s D_{22}^{sh} \neq 0$ when $D_{22}^{sh} \neq 0$ for the h th fault mode to guarantee that $e_{22\epsilon}^s$ is sensitive to $D_{22}^{sh} \Delta_1^s(\cdot)$.

Accordingly, the $\Psi^s(\cdot)$ in (29) is given by

$$\begin{aligned} \Psi^s(\cdot) = & T^s A_{211}^s \hat{z}_1 + T^s A_{22}^{s11} z_{21}^s + T^s A_{22}^{s12} z_{22}^s \\ & + T^s g_{21}^s((T_a^s)^{-1} z^s) - \hat{A}_{22}^{s22} T^s z_{21}^s. \end{aligned} \quad (45)$$

For the s th fault mode, when $e_{22}^s \geq \epsilon$, $\dot{e}_{22\epsilon}^s = \dot{e}_{22}^s + T^s \dot{z}_{21}^s$. By substituting (45) to (29) and comparing with (25), the error dynamic is obtained by

$$\begin{aligned} \dot{e}_{22\epsilon}^s = & (A_{212}^s + T^s A_{211}^s) e_1 + \hat{A}_{22}^{s22} e_{22\epsilon}^s \\ & + (g_{22}^s + T^s g_{21}^s)((T_a^s)^{-1} (z^s - \hat{z}^s)) \\ & + (\eta_{22}^s + T^s \eta_{21}^s)(\cdot) + D_{23}^{ss} \Delta_2^s(\cdot, \theta^{ss2}) \\ & - D_{23}^{ss} \Delta_2^s(\cdot, \hat{\theta}^{ss2}) - D_{23}^{ss} \bar{\Lambda}_2^s(\cdot, \hat{\theta}^{ss2}). \end{aligned} \quad (46)$$

Also when $e_{22}^s \leq -\epsilon$, $\dot{e}_{22\epsilon}^s = \dot{e}_{22}^s + T^s \dot{z}_{21}^s$,

$$\begin{aligned} \dot{e}_{22\epsilon}^s = & (A_{212}^s + T^s A_{211}^s) e_1 + \hat{A}_{22}^{s22} e_{22\epsilon}^s \\ & + (g_{22}^s + T^s g_{21}^s)((T_a^s)^{-1} (z^s - \hat{z}^s)) \\ & + (\eta_{22}^s + T^s \eta_{21}^s)(\cdot) + D_{23}^{ss} \Delta_2^s(\cdot, \theta^{ss2}) \\ & - D_{23}^{ss} \Delta_2^s(\cdot, \hat{\theta}^{ss2}) + D_{23}^{ss} \bar{\Lambda}_2^s(\cdot, \hat{\theta}^{ss2}). \end{aligned} \quad (47)$$

Remark 5. The estimation error $\text{col}(e_{21}^s, e_{22}^s)$ is chosen as isolation residual in [13] and [14] while only $e_{22\epsilon}^s$ is selected as residual in this paper, which and the characteristic that $e_{21}^s = 0$ when sliding motion occurs facilitate the design of more proper adaptive thresholds to improve isolability. ∇

According to (46) and (47), the adaptive threshold interval is defined as $[\underline{\zeta}^s, \bar{\zeta}^s]$ where

$$\dot{\zeta}^s = (\Upsilon_1 + \bar{\Upsilon}_2) \mathcal{I}_{q_s} + \hat{A}_{22}^{s22} \bar{\zeta}^s, \quad \dot{\bar{\zeta}}^s = -(\Upsilon_1 + \underline{\Upsilon}_2) \mathcal{I}_{q_s} + \hat{A}_{22}^{s22} \underline{\zeta}^s \quad (48)$$

where $\Upsilon_1 = (\|A_{212}^s + T^s A_{211}^s\| + \mathcal{L}(1 + \|T_0^s\|)) \chi(t) + (1 + \|T^s\|) \bar{\eta} > 0$, $\bar{\Upsilon}_2 = \|\bar{W}_2^s(\cdot, \hat{\theta}^{ss2})\| \|\hat{\theta}^{ss2}\| > 0$ and $\underline{\Upsilon}_2 = \|\underline{W}_2^s(\cdot, \hat{\theta}^{ss2})\| \|\hat{\theta}^{ss2}\| > 0$.

Under the initial condition that $\bar{\zeta}^s(T_d) \geq 0$ and $\underline{\zeta}^s(T_d) \leq 0$, it has that $\bar{\zeta}^s(t) \geq 0$ and $\underline{\zeta}^s(t) \leq 0$, $\forall t > T_d$ because \hat{A}_{22}^{s22} is Metzler.

Define two errors $\bar{e}_{22\epsilon}^s = \bar{\zeta}^s - e_{22\epsilon}^s$ when $e_{22}^s \geq \epsilon$, and $\underline{e}_{22\epsilon}^s = e_{22\epsilon}^s - \underline{\zeta}^s$ when $e_{22}^s \leq -\epsilon$, respectively. Assume that

$\underline{\zeta}^s(T_d) \leq e_{22\epsilon}^s(T_d) \leq \bar{\zeta}^s(T_d)$, $\bar{e}_{22\epsilon}^s(T_d) \geq 0$ and $\underline{e}_{22\epsilon}^s(T_d) \geq 0$. Then

$$\dot{\bar{e}}_{22\epsilon}^s = \hat{A}_{22}^{s22} \bar{e}_{22\epsilon}^s + \bar{\phi}, \quad \bar{e}_{22\epsilon}^s(T_d) \geq 0, \quad (49)$$

$$\dot{\underline{e}}_{22\epsilon}^s = \hat{A}_{22}^{s22} \underline{e}_{22\epsilon}^s + \underline{\phi}, \quad \underline{e}_{22\epsilon}^s(T_d) \geq 0 \quad (50)$$

where $\bar{\phi}$ and $\underline{\phi}$ can be obtained based on (46), (47) and (48).

From (71) and (72) in Appendix with $\alpha = e_{22}^s$, it can be obtained that when $e_{22}^s \geq \epsilon$, $\bar{\phi} \geq 0$, and when $e_{22}^s \leq -\epsilon$, $\underline{\phi} \geq 0$.

Consider when one of the component $\bar{e}_{22\epsilon}^s$ of $\bar{e}_{22\epsilon}^s$ in (49) is equal to zero for the first time at $t = t_1 > T_d$. Then,

$$\dot{\bar{e}}_{22\epsilon}^s(t_1) = \sum_{j=1}^{q_s} \hat{A}_{22ij}^{s22} \bar{e}_{22\epsilon}^s(t_1) + \bar{\phi}_j \quad (51)$$

where \hat{A}_{22ij}^{s22} represent the i th row and j th column of \hat{A}_{22}^{s22} . At $t = t_1$, $\bar{e}_{22\epsilon}^s(t_1) = 0$, $\bar{e}_{22\epsilon}^s(t_1) \geq 0$, $j \neq i$ and $\bar{\phi}_j \geq 0$. From the fact that \hat{A}_{22}^{s22} is Metzler, $\hat{A}_{22ij}^{s22} > 0$, $i \neq j$, then $\dot{\bar{e}}_{22\epsilon}^s(t_1) \geq 0$, which implies that $\bar{e}_{22\epsilon}^s$ will stay nonnegative. Finally $\bar{e}_{22\epsilon}^s$ remains nonnegative for any time $t \geq T_d$. Therefore, $\bar{e}_{22\epsilon}^s \geq 0, \forall t > T_d$. Using the same analysis, the result that $\underline{e}_{22\epsilon}^s \geq 0, \forall t > T_d$ can be obtained.

Then, the following proposition is ready to be presented.

Proposition 2. If $0 \leq e_{22\epsilon}^s(T_d) \leq \bar{\zeta}^s(T_d)$, then $\bar{\zeta}^s(t) \geq 0$, and $e_{22\epsilon}^s(t) \leq \bar{\zeta}^s(t)$, $\forall t > T_d$, else if $\underline{\zeta}^s(T_d) \leq e_{22\epsilon}^s(T_d) \leq 0$, $\underline{\zeta}^s(t) \leq 0$, $\underline{\zeta}^s(t) \leq e_{22\epsilon}^s(t)$, $\forall t > T_d$. Furthermore, if $\underline{\zeta}^s(T_d) \leq e_{22\epsilon}^s(T_d) \leq \bar{\zeta}^s(T_d)$, then $\underline{\zeta}^s(t) \leq e_{22\epsilon}^s(t) \leq \bar{\zeta}^s(t)$, $\forall t > T_d$.

Proof: The result is obtained directly from the analysis above and the proof is omitted here. ■

It should be noted that the estimation error $\tilde{\theta}^{ss2}$ is used in (48). Since θ^{ss2} is unknown, $\bar{\zeta}^s$ and $\underline{\zeta}^s$ cannot be used in the design directly. The projection adaptive law proposed in (32) ensures that $\|\tilde{\theta}^{ss2}\|$ is bounded by a known constant κ_1 . Therefore, in the fault isolation scheme, $\|\tilde{\theta}^{ss2}\|$ in (48) can be replaced by κ_1 in practical design.

D. Incipient Sensor Fault Isolability Analysis

For the r th FIE ($r = 1, \dots, N$), there exist two variables $\bar{\vartheta}^{rr}$ and $\underline{\vartheta}^{rr}$ satisfying that

$$\dot{\bar{\vartheta}}^{rr} = \hat{A}_{22}^{r22} \bar{\vartheta}^{rr} + D_{23}^{rr} \left(\Delta_2^r(\cdot, \hat{\theta}^{rr2}) + \bar{\Lambda}_2^r(\cdot, \hat{\theta}^{rr2}) \right), \quad (52)$$

$$\dot{\underline{\vartheta}}^{rr} = \hat{A}_{22}^{r22} \underline{\vartheta}^{rr} + D_{23}^{rr} \left(\Delta_2^r(\cdot, \hat{\theta}^{rr2}) + \underline{\Lambda}_2^r(\cdot, \hat{\theta}^{rr2}) \right) \quad (53)$$

where $\bar{\vartheta}^{rr}(T_d) = 0$ and $\underline{\vartheta}^{rr}(T_d) = 0$. Also for the s th fault mode and the r th FIE, there exists a variable ζ^{rs} satisfying that $\zeta^{rs}(T_d) = 0$,

$$\dot{\zeta}^{rs} = \hat{A}_{22}^{r22} \zeta^{rs} + T^r D_{22}^{rs} \Delta_1^r(\cdot, \theta^{rs1}) + D_{23}^{rs} \Delta_2^r(\cdot, \theta^{rs2}). \quad (54)$$

To measure the different effects between the two fault modes, the two functions between the s th fault mode and the h th fault mode are defined by

$$\bar{J}^{rs} = \zeta^{rs} - \bar{\vartheta}^{rr}, \quad \underline{J}^{rs} = \zeta^{rs} - \underline{\vartheta}^{rr}. \quad (55)$$

Remark 6. From a qualitative point of view, \bar{J}^{rs} and \underline{J}^{rs} can be interpreted as a filtered version of the difference between the actual fault functions $T^r D_{22}^{rs} \Delta_1^r(y, u, \theta^{rs1}) +$

$D_{23}^{rs} \Delta_2^r(y, u, \theta^{rs2})$ and its estimation $D_{23}^{rs} \Delta_2^s(y, u, \hat{\theta}^{rs2})$ associated with the r th FIE whose structure does not match the actual s th fault mode. The functions given in (55), defined as the ability of the r th FIE to match the s th fault mode, provide a quantitative measure of the difference between the s th fault mode and the r th fault mode. ▽

Then the following theorem is ready to be presented.

Theorem 1. Consider the FIEs described by (27)-(29). Suppose that the s th fault mode occurs at time $t = T_0$ which is detected at time $t = T_d$. The s th fault mode is isolable if for each $r \in \{1, \dots, N\} \setminus \{s\}$, there exist certain time $t^r > T_d$ and some $j \in \{1, \dots, q_r\}$ such that the functions \bar{J}_j^{rs} or \underline{J}_j^{rs} satisfy that

$$\bar{J}_j^{rs}(t^r) \leq -\bar{\mathcal{F}}_j^{rs} - \bar{\delta}_j^r, \quad \underline{J}_j^{rs}(t^r) \geq \underline{\mathcal{F}}_j^{rs} - \underline{\zeta}_j^r \quad (56)$$

where \bar{J}_j^{rs} and \underline{J}_j^{rs} are the j th component of \bar{J}^{rs} and \underline{J}^{rs} respectively, $\bar{\delta}_j^r$ and $\underline{\zeta}_j^r$ are the j th component of $\bar{\zeta}^r$ and $\underline{\zeta}^r$ respectively, $\bar{\mathcal{F}}_j^{rs}$ and $\underline{\mathcal{F}}_j^{rs}$ will be given later.

Proof: When the s th fault mode occurs, the r th FIE dynamic is described by

$$\begin{aligned} \dot{e}_{22}^r = & A_{212}^r e_1 + \hat{A}_{22}^{r22} \left(e_{22}^r - \epsilon \text{sat}\left(\frac{e_{22}^r}{\epsilon}\right) \right) + g_{22}^r \left((T_a^r)^{-1} z^r \right) \\ & - g_{22}^r \left((T_a^r)^{-1} \hat{z}^r \right) + \eta_{22}^r(\cdot) + D_{23}^{rs} \Delta_2^r(\cdot, \theta^{rs2}) \\ & - D_{23}^{rr} \Delta_2^r(\cdot, \hat{\theta}^{rr2}) - D_{23}^{rr} \Lambda_2^r(\cdot, \hat{\theta}^{rr2}) \text{sat}\left(\frac{e_{22}^r}{\epsilon}\right) - \Psi(\cdot). \end{aligned} \quad (57)$$

From the definition of the residual $e_{22\epsilon}^r = e_{22}^r - \epsilon \text{sat}\left(\frac{e_{22}^r}{\epsilon}\right) + T^r z_{21}^r$, the dynamics of $e_{22\epsilon}^r$ can be obtained based on (57).

Let $\bar{e}_{22\epsilon}^{rs} = e_{22\epsilon}^r + \bar{\vartheta}^{rr} - \zeta^{rs}$. It follows from the dynamics of $e_{22\epsilon}^r$ that

$$\begin{aligned} \dot{\bar{e}}_{22\epsilon}^{rs} = & (A_{212}^r + T^r A_{211}) e_1 + \hat{A}_{22}^{r22} \bar{e}_{22\epsilon}^{rs} + (T^r g_{21}^r + g_{22}^r) \\ & \left((T_a^r)^{-1} z^r - (T_a^r)^{-1} \hat{z}^r \right) + (T^r \eta_{21} + \eta_{22}^r)(\cdot). \end{aligned} \quad (58)$$

Since \hat{A}_{22}^{r22} is Hurwitz, using the Bellman-Gronwall lemma and the similar reasoning as that used in (37), there exists a bank of positive time functions $\bar{\mathcal{F}}_j^{rs}(t)$, $j = 1, \dots, q_s$ such that

$$-\bar{\mathcal{F}}_j^{rs}(t) \leq \bar{e}_{22\epsilon}^{rs} \leq \bar{\mathcal{F}}_j^{rs}(t) \quad (59)$$

where $\bar{e}_{22\epsilon}^{rs}$ represents the j th component of $\bar{e}_{22\epsilon}^{rs}$ in (58).

Let $\underline{e}_{22\epsilon}^{rs} = e_{22\epsilon}^r + \underline{\vartheta}^{rr} - \zeta^{rs}$ when $e_{22}^r \leq -\epsilon$. Similar with (59), there exists a bank of positive time functions $\underline{\mathcal{F}}_j^{rs}(t)$ such that

$$-\underline{\mathcal{F}}_j^{rs}(t) \leq \underline{e}_{22\epsilon}^{rs} \leq \underline{\mathcal{F}}_j^{rs}(t) \quad (60)$$

where $\underline{e}_{22\epsilon}^{rs}$ represents the j th component of $\underline{e}_{22\epsilon}^{rs}$.

To isolate the s th fault mode, it requires that at least one component $e_{22\epsilon}^{rs}$, $j = 1, \dots, q_r$ of $e_{22\epsilon}^r$, $r \in \{1, \dots, N\} \setminus \{s\}$ runs out of the adaptive threshold interval $[\underline{\zeta}_j^r, \bar{\zeta}_j^r]$. Since, when $e_{22\epsilon}^r > 0$, $e_{22\epsilon}^r = \bar{e}_{22\epsilon}^{rs} - \bar{J}^{rs}$, and when $e_{22\epsilon}^r < 0$, $e_{22\epsilon}^r = \underline{e}_{22\epsilon}^{rs} - \underline{J}^{rs}$, then

$$e_{22\epsilon}^r = \bar{e}_{22\epsilon}^{rs} - \bar{J}_j^{rs} \geq \bar{\zeta}_j^r \text{ or } e_{22\epsilon}^r = \underline{e}_{22\epsilon}^{rs} - \underline{J}_j^{rs} \leq \underline{\zeta}_j^r. \quad (61)$$

Hence, eqs. (56) are obtained, and the result follows. ■

Remark 7. It can be seen from (56) in Theorem 1 that the matrices A_f^s and A_f^r do not affect isolation ability directly because (56) does not contain A_f^s and A_f^r . From the analysis in Section III-B, it obtains that A_f^i does not affect the stability of sliding mode and adaptive FIEs. Therefore, matrices A_f^i in (4) are chosen by considering not only the practical characteristics of incipient rectifier voltage sensor faults, but also the enlargement of the differences between θ^i . ∇

Remark 8. It is worthy to point out that it is challenging to develop multi sensor faults isolation schemes only using sliding mode technique (see e.g., [9]). In this paper, the nonlinear parameterization adaptive estimation technique is introduced to propose novel multi incipient sensor faults isolation schemes which are quite different from single sensor fault isolation schemes in [14]. ∇

IV. VERIFICATION

In this section, simulation based on the model (6)-(7) and experiment based on TDCS-FIB (traction and driving control system-fault injection benchmark) will be presented. The TDCS-FIB is programmed based on SimPower System toolbox to simulate common faults that may occur on CRH2 by Central South University Fault Injection Team, which is outside the red rectangle frame in Fig. 2. The designed incipient voltage sensor fault isolation schemes are included in the red rectangle frame in Fig. 2. In both simulation and exper-

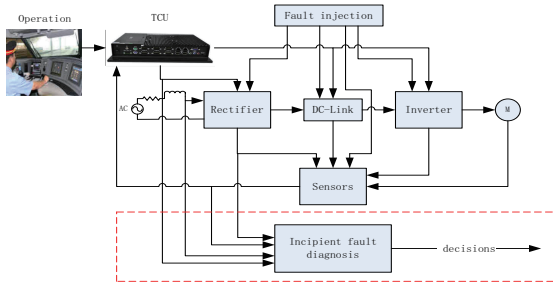


Fig. 2. Schematic diagrams of TDCS-FIB and fault isolation schemes.

iment, the railway is supposed to work at a fixed velocity. Then the instantaneous power P_m is fixed. The parameters in model (6)-(7) and values of the electrical components in the rectifier device in Fig. 2 are given by Table I. Thus, the matrices

TABLE I
PARAMETERS.

Parameter	Value	Unit
P_m	800	kW
R	0.34	Ω
L	2.2×10^{-3}	H
C_1	1.6×10^{-3}	F
C_2	1.6×10^{-3}	F
u_s	$1500\sqrt{2}\sin(314t)$	V

in (6)-(7) are given by $A_n = \begin{bmatrix} -154.55 & -454.55S_1 & 454.55S_2 \\ 62.50S_1 & 0 & 0 \\ -62.50S_2 & 0 & 0 \end{bmatrix}$, $B_n = \begin{bmatrix} 454.55 \\ 0 \\ 0 \end{bmatrix}$ and $g_n(X) = \begin{bmatrix} 0 \\ -\frac{5 \times 10^6}{X_2 + X_3} \\ -\frac{5 \times 10^6}{X_2 + X_3} \end{bmatrix}$ where $S_i, i = 1, 2$ are generated by TCU (traction control unit) module.

Consider the 3-order harmonic of grid side current i_s and parameter uncertainties $\Delta_R = 0.02\Omega$, $\Delta_L = 0.2 \times 10^{-4}H$ and $\Delta_C = 2 \times 10^{-4}F$ in this paper. Then $\eta_{i_s} = \frac{H_{i_s} i_s}{\sqrt{2} \cos(\omega t)} \sin(3\omega t - \varphi) + \Delta_{1i_s} i_s + \Delta_{2i_s} u_1 + \Delta_{3i_s} u_2$, $\eta_{u_1} = \Delta_{1u_1} i_s$ and $\eta_{u_2} = \Delta_{1u_2} i_s$ with $H_{i_s} = 100$, $\omega = 314$, $\Delta_{1i_s} = 7.6145$, $\Delta_{2i_s} = 4.0950$, $\Delta_{3i_s} = -4.0950$, $\Delta_{1u_1} = -0.7716$ and $\Delta_{1u_2} = 0.7716$. Thus, $\bar{\eta}$ in Assumption 1 is obtained by

$$\bar{\eta} \geq \left\| \begin{bmatrix} \frac{H_{i_s} \sin(3\omega t - \varphi)}{\sqrt{2} \cos(\omega t)} + \Delta_{1i_s} & \Delta_{2i_s} & \Delta_{3i_s} \\ \Delta_{1u_1} & 0 & 0 \\ \Delta_{1u_2} & 0 & 0 \end{bmatrix} \right\| \|X\|.$$

Note that $i_s \in [0, 500]$ and $u_1, u_2 \in [1000, 1500]$ in CRH2. Then the constant bound $\bar{\eta}$ can be obtained. Using differential mean value theorem presented in [26], the Lipschitz constant of $g_n(X)$ in Assumption 1 is $\mathcal{L} = 1.58$.

The base function $\delta(\cdot)$ used to construct $\Delta(\cdot)$ in (8) is given by

$$\delta(\cdot) = \zeta(u_1, u_2, t)\theta^2 \quad (62)$$

where $\zeta(u_1, u_2, t) = 0.01(u_1 \sin(314t) + u_2 \sin(314t + \pi/3) + (u_1 + u_2) \sin(314t + 2\pi/3))$. The distribution matrices in (9) and the basis function $\delta(\cdot)$ are all used in the sequel simulation and experiment.

A. Simulation

In the 1st fault mode, the incipient fault is modeled by

$$\dot{f}^1 = A_f^1 f^1 + D_2^1 \Delta(u_1, u_2, \theta^1), \quad f^1(0) = 0 \quad (63)$$

where $A_f^1 = -100I_3$. It is assumed that fault parameters θ_1^1 , θ_2^1 and θ_3^1 in $\Delta(\cdot, \theta^1) = \text{col}(\delta(\cdot, \theta_1^1), \delta(\cdot, \theta_2^1), \delta(\cdot, \theta_3^1))$ belong to the intervals that $\theta_1^1 = 0$, $\theta_2^1 \in [0, 100]$, $\theta_3^1 = 0$. In the 2nd fault mode, the incipient faults is modeled by

$$\dot{f}^2 = A_f^2 f^2 + D_2^2 \Delta(u_1, u_2, \theta^2), \quad f^2(0) = 0 \quad (64)$$

where $A_f^2 = -20I_3$ and the intervals of fault parameters θ_1^2 , θ_2^2 and θ_3^2 in $\Delta(\cdot, \theta^2)$ are given by $\theta_1^2 = 0$, $\theta_2^2 = 0$ and $\theta_3^2 \in [0, 50]$. In the 3rd fault mode, the incipient faults is modeled by

$$\dot{f}^3 = A_f^3 f^3 + D_2^3 \Delta(u_1, u_2, \theta^3), \quad f^3(0) = 0 \quad (65)$$

where $A_f^3 = -50I_3$ and the intervals of fault parameters θ_1^3 , θ_2^3 and θ_3^3 in $\Delta(\cdot, \theta^3)$ are given by $\theta_1^3 = 0$, $\theta_2^3 \in [0, 60]$ and $\theta_3^3 \in [0, 80]$.

Suppose that the fault parameters $\theta_1^3 = 0, \theta_2^3 = 0, \theta_3^3 = 0$ before $0.15s$, and $\theta_1^3 = 0, \theta_2^3 = 40, \theta_3^3 = 60$ after $0.15s$. As comparison, the fault isolation method in [13] and [14] is used to isolate this incipient sensor fault mode firstly. The simulation results of the 1st FIE constructed based on [13] and [14] are shown in Figs. 3 and 4. Since that It can be seen from Fig. 3 that the incipient fault parameter estimation lines are not convergent, which means that the adaptive laws in [13] and [14] can not be used to estimate incipient faults parameterized nonlinearly efficiently. Furthermore, it is clear from Figs. 4 that none of the isolation residuals exceeds their adaptive thresholds, which is conflicting with the isolation requirement in [13] and [14]. Thus, there is no need to show the simulation results of the 2ed FIE and no isolation decision can be made based on the principles provided in [13] and [14].

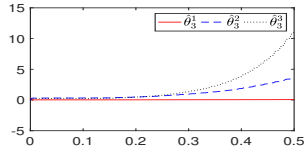


Fig. 3. Parameter estimation lines in the 1st and 2nd FIEs.

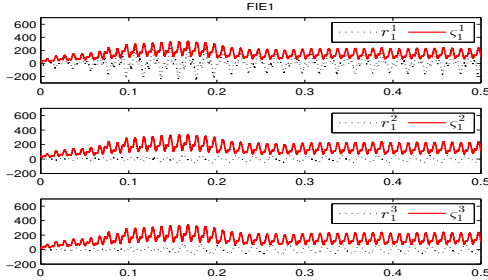


Fig. 4. Residuals of the 1rd FIE (dashed and black lines) and corresponding adaptive thresholds (solid and red lines).

The reason is that multi sensor faults cases are not considered in these two paper.

Then, the multi incipient faults isolation schemes developed in this paper are constructed. Corresponding to the three fault modes, there are three groups coordinate transformation matrices for three FIEs. The 1st group is described by

$$T_a^1 = \begin{bmatrix} I_3 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 1 \end{bmatrix}, T_b^1 = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & 1 & 0 \end{bmatrix},$$

For the 2nd and 3rd groups, $T_a^2 = T_a^3 = I_6$ and $T_b^2 = T_b^3 = I_3$. Two important schemes in FIEs are constructing adaptive laws $\hat{\theta}^{ss1}$ and $\hat{\theta}^{ss2}$ for nonlinear incipient fault parameters in (31) and (32), and selecting gains $m_{21}^s(\cdot)$ and $M_{22}^s(\cdot)$ to ensure sliding motion in Proposition 1. In two schemes, $\lambda_i(\cdot)$, $\bar{w}_i(\cdot)$ and $w_i(\cdot)$, $i = 1, 2, 3$ are used to construct $W_1^i(\cdot)$, $\Lambda_1^i(\cdot)$, $W_2^i(\cdot)$ and $\Lambda_2^i(\cdot)$. Based on Lemma 2, these terms for the 3rd incipient fault mode (65) with nonlinear parameterization (62) in the 1st FIE are expressed by $\bar{\lambda}_1(\cdot) = 0$, $\bar{\lambda}_2(\cdot) = 0$, $\bar{\lambda}_3(\cdot) = 0$, $\bar{w}_1(\cdot) = 0$, $\bar{w}_2(\cdot) = -100\delta(\cdot)$, $\bar{w}_3(\cdot) = -100\delta(\cdot)$, $\underline{\lambda}_1(\cdot) = 0$, $\underline{\lambda}_2(\cdot) = -\delta(\cdot)(\hat{\theta}_3^3)^2 + 100\delta(\cdot)\hat{\theta}_3^3$, $\underline{\lambda}_3(\cdot) = -\delta(\cdot)(\hat{\theta}_2^3)^2 + 100\delta(\cdot)\hat{\theta}_2^3$, $w_1(\cdot) = 0$, $w_2(\cdot) = -2\delta(\cdot)\hat{\theta}_3^3$ and $w_3(\cdot) = -2\delta(\cdot)\hat{\theta}_2^3$. The expressions of those terms in the 2nd and the 3rd FIEs are similar to these in the 1st FIE, which are omitted here due to space limitation. Then the adaptive laws in (31) and (32), and gains in (39) and (40) can be constructed. Thus, based on (27)-(29), these three FIEs corresponding to the three incipient sensor fault modes can be easily constructed.

The estimation lines of incipient fault parameters in three developed FIEs are shown in Fig. 5. It can be seen that all the estimations lines are bounded. The incipient sensor fault isolation results are shown in Figs. 6 and 7. It can be seen from Fig. 6 that r^1 in the 1st FIE exceeds the lower bound $\underline{\varsigma}^1$ at t_1 , and r^2 in the 2st FIE exceeds $\underline{\varsigma}^2$ at t_2 as well, while r_1^3 and r_2^3 are both stay in their threshold intervals $[\underline{\varsigma}_1^3, \bar{\varsigma}_1^3]$ and $[\underline{\varsigma}_2^3, \bar{\varsigma}_2^3]$ all the time respectively in Fig. 7. As a result, based

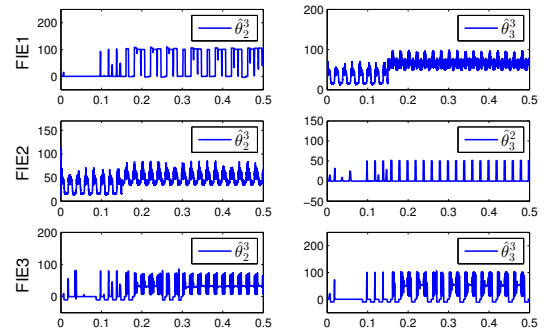


Fig. 5. Parameter estimation lines in the 1st, 2nd and 3rd FIEs.

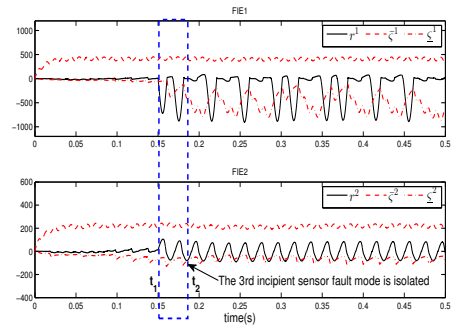


Fig. 6. Residuals of the 1st and 2nd FIE (solid and black lines) and corresponding adaptive threshold intervals (dashed and red lines).

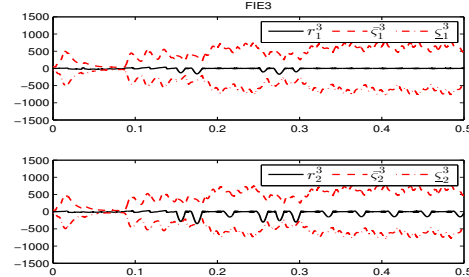


Fig. 7. Residuals of the 3rd FIE (solid and black lines) and corresponding adaptive threshold intervals (dashed and red lines).

on the isolation principle developed in this paper, the decision that the 3rd incipient sensor fault mode occurs at time instant t_2 is made.

B. Experiment

The TDCS-FIB in Fig. 2 can effectively simulate some common faults occurring in the electric traction and driving system of high-speed railway, which provides a good platform to study the fault diagnosis issue for electrical traction systems. The two voltage sensors in DC-Link module in Fig. 2 are both injected incipient faults provided by fault expression (65) with incipient sensor fault parameters θ_1^3 , θ_2^3 and θ_3^3 given as in the simulation part. The developed incipient sensor fault isolation schemes are established and added to the traction system in the way presented in the red rectangle frame in Fig. 2.

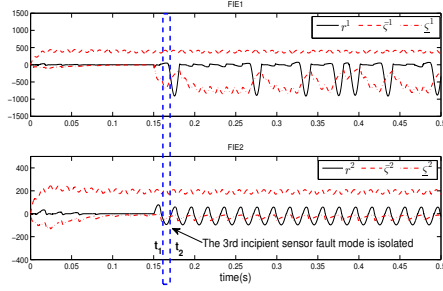


Fig. 8. Residuals of the 1st and 2nd FIE (solid and black lines) and corresponding adaptive threshold intervals (dashed and red lines).

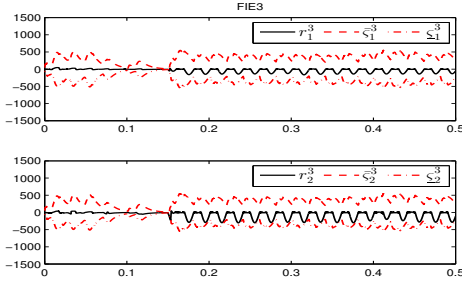


Fig. 9. Residuals of the 3rd FIE (solid and black lines) and corresponding adaptive threshold intervals (dashed and red lines).

The response curves in experiment in Fig. 8 and Fig. 9 are similar with those in simulation in Figs. 6 and 7. It can be seen from Fig. 8 and Fig. 9 that r^2 in the 2nd FIE exceeds $[\zeta^2, \bar{\zeta}^2]$ at t_1 , and r^1 in the 1st FIE exceeds $[\zeta^1, \bar{\zeta}^1]$ at t_2 . However, residuals $r_1^3 \in [\zeta_1^3, \bar{\zeta}_1^3]$ and $r_2^3 \in [\zeta_2^3, \bar{\zeta}_2^3]$ all the time. Therefore, based on the isolation principle developed in this paper, it can make the decision that both voltage sensors have incipient faults after time instant t_2 .

V. CONCLUSION

This paper has proposed a dc voltage incipient sensor fault isolation scheme for single-phase three-level rectifier devices in high-speed railway electrical traction systems. A novel incipient fault isolation method has been developed by combining sliding mode technique with nonlinear parametrization adaptive estimation technique. In the proposed method, novel residuals and adaptive threshold intervals have been presented to isolate different fault modes. The isolability has been studied and the sufficient isolable conditions have been derived by the defined functions. Based on TDCS-FIB, simulation and experiment results have demonstrated that the proposed method is effective and practicable.

APPENDIX

Lemma 2. For any bounded continuous known scalar function $f(\phi, \theta)$ with ϕ being known and $\theta \in \Theta \subset \mathcal{R}^h$, and any given $\hat{\theta} \in \Theta$, there exist vector function $w(\phi, \hat{\theta}) \in \mathcal{R}^h$ and scalar function $\lambda(\phi, \hat{\theta})$ such that

$$J(w(\cdot), \theta) - \lambda(\cdot) \leq 0 \quad (66)$$

where $J(w(\cdot), \theta) = \text{sign}(\alpha) (f(\phi, \theta) - f(\phi, \hat{\theta}) + w(\cdot)(\theta - \hat{\theta}))$ with α being known.

Proof: Refer [19] and [24]. \blacksquare

According to Lemma 2, for any $\delta(\cdot, \theta_r^j)$, $\theta_r^j \in \mathcal{R}^h$, $r = 1, \dots, p$, $j = 1, \dots, N$, there exist row function vector $\underline{w}_r(\cdot, \hat{\theta}_r^j) \in \mathcal{R}^h$ and scalar function $\underline{\lambda}_r(\cdot, \hat{\theta}_r^j)$ such that

$$\delta(\cdot, \theta_r^j) - \delta(\cdot, \hat{\theta}_r^j) + \underline{w}_r(\cdot, \hat{\theta}_r^j)(\theta_r^j - \hat{\theta}_r^j) + \underline{\lambda}_r(\cdot, \hat{\theta}_r^j) > 0, \quad (67)$$

and row function vector $\bar{w}_r(\cdot, \hat{\theta}_r^j) \in \mathcal{R}^h$ and scalar function $\bar{\lambda}_r(\cdot, \hat{\theta}_r^j)$ such that

$$\delta(\cdot, \theta_r^j) - \delta(\cdot, \hat{\theta}_r^j) + \bar{w}_r(\cdot, \hat{\theta}_r^j)(\theta_r^j - \hat{\theta}_r^j) - \bar{\lambda}_r(\cdot, \hat{\theta}_r^j) < 0. \quad (68)$$

Given $\hat{\theta}^{ij1} \in \Theta^{ij1}$, for $\Delta_1^i(\cdot, \theta^{ij1})$ in (19), there exist diagonal matrix functions $\bar{W}_1^i(\cdot, \hat{\theta}^{ij1})$ and $\underline{W}_1^i(\cdot, \hat{\theta}^{ij1})$ and vector functions $\bar{\Lambda}_1^i(\cdot, \hat{\theta}^{ij1})$ and $\underline{\Lambda}_1^i(\cdot, \hat{\theta}^{ij1})$ such that

$$\Delta_1^s(\cdot, \theta^{ij1}) - \Delta_1^s(\cdot, \hat{\theta}^{ij1}) + \underline{W}_1^s(\cdot)(\theta^{ij1} - \hat{\theta}^{ij1}) + \underline{\Lambda}_1^s(\cdot) > 0, \quad (69)$$

$$\Delta_1^s(\cdot, \theta^{ij1}) - \Delta_1^s(\cdot, \hat{\theta}^{ij1}) + \bar{W}_1^s(\cdot)(\theta^{ij1} - \hat{\theta}^{ij1}) - \bar{\Lambda}_1^s(\cdot) < 0 \quad (70)$$

where $\bar{W}_1^i(\cdot, \hat{\theta}^{ij1}) = \text{diag}(\bar{w}_r(\cdot, \hat{\theta}^{ij1}))$, $\underline{W}_1^i(\cdot, \hat{\theta}^{ij1}) = \text{diag}(\underline{w}_r(\cdot, \hat{\theta}^{ij1}))$, $\bar{\Lambda}_1^i(\cdot, \hat{\theta}^{ij1}) = \text{col}(\bar{\lambda}_r(\cdot, \hat{\theta}^{ij1}))$ and $\underline{\Lambda}_1^i(\cdot, \hat{\theta}^{ij1}) = \text{col}(\underline{\lambda}_r(\cdot, \hat{\theta}^{ij1}))$, $r = 1, \dots, p - q_i$.

Also, given $\hat{\theta}^{ij2} \in \Theta^{ij2}$ for $\Delta_2^i(\cdot, \theta^{ij2})$ in (21), there exist diagonal matrix functions $\bar{W}_2^i(\cdot, \hat{\theta}^{ij2})$ and $\underline{W}_2^i(\cdot, \hat{\theta}^{ij2})$ and vector functions $\bar{\Lambda}_2^i(\cdot, \hat{\theta}^{ij2})$ and $\underline{\Lambda}_2^i(\cdot, \hat{\theta}^{ij2})$ such that

$$\Delta_2^i(\cdot, \theta^{ij2}) - \Delta_2^i(\cdot, \hat{\theta}^{ij2}) + \underline{W}_2^i(\cdot)(\theta^{ij2} - \hat{\theta}^{ij2}) + \underline{\Lambda}_2^i(\cdot) > 0, \quad (71)$$

$$\Delta_2^i(\cdot, \theta^{ij2}) - \Delta_2^i(\cdot, \hat{\theta}^{ij2}) + \bar{W}_2^i(\cdot)(\theta^{ij2} - \hat{\theta}^{ij2}) - \bar{\Lambda}_2^i(\cdot) < 0 \quad (72)$$

where $\bar{W}_2^i = \text{diag}(\bar{w}_r(\cdot, \hat{\theta}^{ij2}))$, $\underline{W}_2^i = \text{diag}(\underline{w}_r(\cdot, \hat{\theta}^{ij2}))$, $\bar{\Lambda}_2^i = \text{col}(\bar{\lambda}_r(\cdot, \hat{\theta}^{ij2}))$ and $\underline{\Lambda}_2^i = \text{col}(\underline{\lambda}_r(\cdot, \hat{\theta}^{ij2}))$, $r = p - q_i + 1, \dots, p$. The $\bar{W}_1^i(\cdot, \hat{\theta}^{ij1})$ and $\underline{W}_1^i(\cdot, \hat{\theta}^{ij1})$ are defined as

$$\begin{cases} \bar{W}_1^i(\cdot, \cdot) = \bar{W}_1^i(\cdot, \cdot), \underline{\Lambda}_1^i(\cdot, \cdot) = \bar{\Lambda}_1^i(\cdot, \cdot), \alpha > 0, \\ \underline{W}_1^i(\cdot, \cdot) = \underline{W}_1^i(\cdot, \cdot), \underline{\Lambda}_1^i(\cdot, \cdot) = \underline{\Lambda}_1^i(\cdot, \cdot), \alpha < 0, \end{cases} \quad (73)$$

and $\bar{W}_2^i(\cdot, \hat{\theta}^{ij2})$ and $\underline{W}_2^i(\cdot, \hat{\theta}^{ij2})$ are defined as

$$\begin{cases} \bar{W}_2^i(\cdot, \cdot) = \bar{W}_2^i(\cdot, \cdot), \underline{\Lambda}_2^i(\cdot, \cdot) = \bar{\Lambda}_2^i(\cdot, \cdot), \alpha > 0, \\ \underline{W}_2^i(\cdot, \cdot) = \underline{W}_2^i(\cdot, \cdot), \underline{\Lambda}_2^i(\cdot, \cdot) = \underline{\Lambda}_2^i(\cdot, \cdot), \alpha < 0. \end{cases} \quad (74)$$

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